# BRITISH

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# VERY LOW FREQUENCY SOUND ATTENUATION MEASUREMENTS

by R. H. Mellen, W. H. Thorp, L. C. Maples, E. N. Jones, D. G. Browning Naval Underwater Systems Center, New London Laboratory U.S.A.

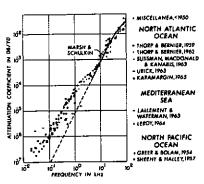


Fig. 1

Measurements of sound propagation in the ocean have established the existence of two distinct regimes of excess attenuation. The first of these has been identified with the MoSO, content of sea water and is indicated by the dashed line of Fig. 1. The

other experimental data below 1 kHz show values some ten times higher, and concur in the existence of an anomaly if not in its precise behavior.

Evidently, the low frequency attenuation anomaly has excited the interest of many investigators who have proposed a number of

### POSSIBLE EXPLANATIONS FOR ATTENUATION ANOMALY

I. SUSPENDED MATTER

2. MOLOGICAL SCATTERING

3. FINITE-AMPLITUDE EFFECTS

4. INHOMOGENEITIES

5. EDDY VISCOSITY

A CHANNEL LEAKAGE

7. INTERNAL WAVES

8. RELAXATION PROCESS (LINKNOWN)

9 PELANATION PROCESS-IONS HORNE, 1968

18. DISSOLVED CO2

12. PLANKTON

DUYKERS, 1967

WESTON, 1966

MAISH, MELLEN, KONTAD, 1965

UBCK, 1963 SCHULKIN, 1963

UECK, 1963

(NO CALCULATIONS) THORP. 1945

URCK, 1966

LE 60Y, 1964

ID. RELAXATION PROCESS-WATER BROWNING, THORF, MELLEN, 1968

FISHER 1969

DUYKEIS 1970

possible explanations (Fig. 2). Many of these

have already been fairly

well discounted and atten-

tion has tended to focus on

a second relaxation-

absorption as a likely

mechanism.

Figure 3 shows the low frequency attenuation data fitted to a relaxation-absorption curve. The complete analytic expression is given by the  $^{\rm ch}$  equation where the first term is the MgSO4 relaxation term ( $f_{\rm r}=64$  kHz) and the second is the anomalous term ( $f_{\rm r}=1$  kHz). The general agreement between continuous wave and explosive measurements is taken as evidence against the finite amplitude explanation.

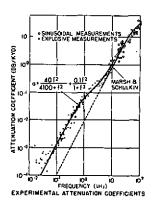


Fig. 3

We have been trying to identify the precise behavior and possibly the cause of the anomaly by carrying out a series of propagation experiments in waters of different temperature and salinity. Figure 4 illustrates our experimental technique. Bodies of water are chosen primarily for their suitability as refraction sound channels so that losses will occur within the water and not at the boundaries. A sound channel is formed by the combined effects of the higher temperature at the surface and the pressure effect at the bottom. The resulting increase in sound

speed toward both boundaries causes the sound to be refracted toward the sound speed minimum which is the sound channel axis.

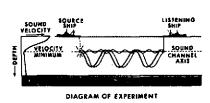


Fig. 4

Explosive charges are detonated on the sound channel axis at various ranges from the listening ship. The acoustic signals are received by means of a hydrophone also located on the axis, and the electrical signal is recorded on magnetic tape for later analysis.

It would seem from the results for the cantilevered triangular plate that it is desirable in such cases to use a minimum number of triangles for greatest accuracy.

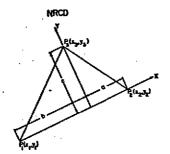
## 4. REFERENCES

- 1. G. R. COWPER et al. 1968 National Research Council of Canada, Aeronautical Rpt. LR-514. A high precision triangular plate-bending element.
- 2. V. MASON 1968 J. Sound Vib. 7, 437. Rectangular finite elements for analysis of plate vibrations.
- 3. P. N. GUSTAFSON et al. 1953 J. Aero. Sci. 20, 331.

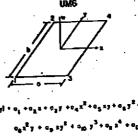
  An experimental study of natural vibrations of cantilevered triangular plates.

	FREQUENCY (Hz.)				
HDDE MUMBA					Experimental Results Reference (3)
1	36.54124	36.33947	36.53897	36.53895	34.3
2	138.9636	138.9567	138,9550	138.9528	136.)
3	193.6010	193.5615	193.3734	193.5699	190.0
	332.7147	332.7060	332.6953	332-6240	325.)
5	453.2197	453.2150	453.0388	452.9050	441.3
•	389.2431	589.1010	589.0678	588.6602	578.0
7	664.0397	663.8512	663-7170	862.9144	
8	798.0966	797.7241	796.9290	796.3389	
9	948.1436	947.2899	946.1370	944.4312	
10	1092.781	1092.583	1091.741	1088,801	

TABLE 1 RATURAL PREQUENCIES OF CANTILEVERED TRIANGULAR PLATE OF ASPECT

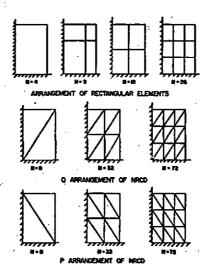


$$w(u,y) = a_1 + a_2 x + a_3 y + a_4 x^2 + a_5 xy + a_6 y^2 + a_7 x^3 + a_6 x^2 y + a_6 xy^2 + a_{12} x^3 y + a_{13} x^2 y^2 + a_{14} x^3 + a_{15} x^3 y^4 + a_{16} x^5 + a_{17} x^3 y^5 + a_{17} x^3 y^4 + a_{18} x^5 + a_{17} x^3 y^5 + a_{17} x^3$$



$$\begin{aligned} & \circ_1 + \circ_2 \mathbf{x} + \circ_3 \mathbf{y} + \circ_4 \mathbf{x}^2 + \circ_5 \mathbf{x} \mathbf{y} + \circ_6 \mathbf{y}^2 + \circ_7 \mathbf{x}^3 \\ & \circ_6 \mathbf{x}^2 \mathbf{y} + \circ_8 \mathbf{x} \mathbf{y}^2 + \circ_{10} \mathbf{y}^3 + \circ_{10} \mathbf{x}^4 + \circ_{10} \mathbf{x}^3 \mathbf{y}^4 \\ & \circ_6 \mathbf{x}^2 \mathbf{y}^2 + \circ_{10} \mathbf{x} \mathbf{y}^3 + \circ_{10} \mathbf{y}^4 + \circ_{10} \mathbf{x}^3 + \circ_{10} \mathbf{x}^3 \mathbf{y}^4 \\ & \circ_{10} \mathbf{y}^3 \mathbf{x}^2 + \circ_{10} \mathbf{y}^3 + \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{x}^4 + \\ & \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{y}^4 + \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{x}^5 + \\ & \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{x}^6 + \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{x}^5 + \\ & \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{x}^6 + \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{x}^5 + \\ & \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{x}^6 + \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{x}^6 + \\ & \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{x}^6 + \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{2}{3} \frac{\mathbf{y}^3}{3} \right) \mathbf{x}^6 + \\ & \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{\mathbf{y}^3}{2} - \frac{\mathbf{y}^3}{2} - \frac{\mathbf{y}^3}{2} - \frac{\mathbf{y}^3}{2} \right) \mathbf{x}^6 + \\ & \circ_{10} \left( \frac{\mathbf{y}^3}{2} - \frac{\mathbf{y}^3}$$

Figure L DISPLACEMENT FUNCTIONS FOR THE UMG AND NRCD ELEMENTS



gure 2. LAYOUT OF FRATE ELEMENTS FOR DYNAMIC ANALYSIS OF THE RECTANGULAR PLATE (NB. only quorier of pions vibrus)

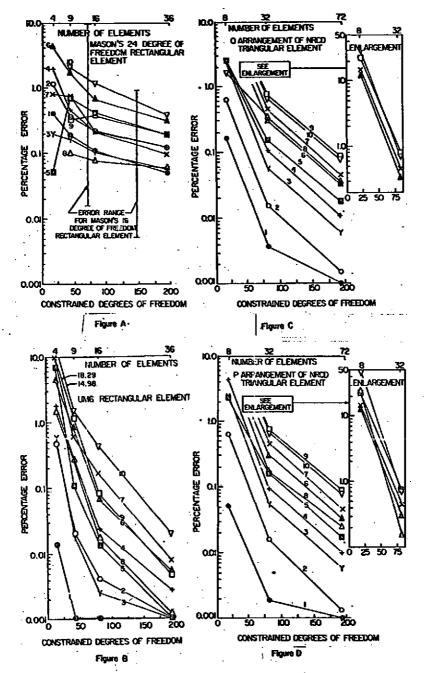


Figure 3. PERCENTAGE ERROR IN THE NATURAL FREQUENCIES OF A 40:27 SIMPLY SUPPORTED PLATE FOR DIFFERENT TRIANGULAR AND RECTANGULAR ELEMENTS.

The recorded signals are analyzed using 1/3-octave filters to determine the peak pressure response at the selected frequencies. The received pressure level is then subtracted from the known source level to obtain the appropriate value of propagation loss as a function of frequency. Figure 5 shows a typical data plot of loss vs. range for the frequency 1410 Hz. From the value of propagation loss 10 log R was subtracted for cylindrical spread-

ing so that the slope of a straight line fit gives the attenuation directly in decibels per unit distance. The intercept at zero range indicates the additional loss due to spherical spreading near the source.

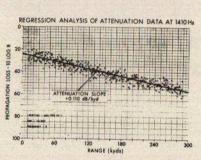


Fig. 5

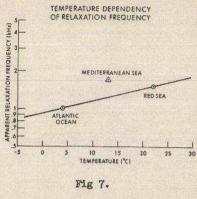
Figure 6 shows our experimental values of d vs. frequency for the 1969 Red Sea Experiment. The solid line is the Thorp curve of Fig. 2. The fit is seen to be reasonably good except for the slightly higher Red Sea value from 1-10 kHz. A more precise relaxation curve fit to the data in fact gives an apparent

RED SEA EXPERIMENTAL ATTENUATION COEFFICIENTS - 1969

Fig. 6

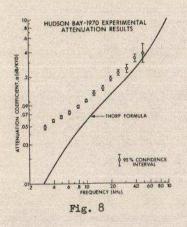
relaxation frequency of 1.5 kHz compared to 1 kHz for Thorp.

The relaxation
hypothesis, of course,
requires an increase in
relaxation frequency with
temperature. Figure 7



shows our two relaxation frequencies vs. temperature together with LeRoy's 1.7 kHz value for the Mediterranean. The lack of a uniform trend of the three values is apparent, even though the overall consistency of relaxation-like behavior of the experimental data is still impressive.

The Hudson's Bay results, Fig. 8, tend to weaken the relaxation hypothesis. (This was a joint U.S./Canadian experiment carried out with Defense Research Establishment Atlantic, Halifax, August 1970.) It is apparent that the values of of are not only considerably in excess of Thorp (solid line) but also do not fit a relaxation curve well at all.



At this point there are several possible reasons for the apparent dilemma: (1) The data may be subject to systematic error (which can be eliminated with improved analytical methods), (2) The simplified sound channel model may be inadequate (we are presently experimenting with a

Fast Fourier Field Program), (3) More than one mechanism may be involved (volume scattering and bottom leakage for example).